



Tribological Performance of Some Pennzane® Based Greases for Vacuum Applications

Mario Marchetti, William R. Jones, Jr., and Kenneth W. Street
Glenn Research Center, Cleveland, Ohio

Donald Wheeler, Duane Dixon, and Mark J. Jansen
Sest, Inc., Middleburg Hts., Ohio

Hiroshi Kimura
Kyodo Yushi Company, Ltd., Kanagawa, Japan

The NASA STI Program Office . . . in Profile

Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) Program Office plays a key part in helping NASA maintain this important role.

The NASA STI Program Office is operated by Langley Research Center, the Lead Center for NASA's scientific and technical information. The NASA STI Program Office provides access to the NASA STI Database, the largest collection of aeronautical and space science STI in the world. The Program Office is also NASA's institutional mechanism for disseminating the results of its research and development activities. These results are published by NASA in the NASA STI Report Series, which includes the following report types:

- **TECHNICAL PUBLICATION.** Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA's counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.
- **TECHNICAL MEMORANDUM.** Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.
- **CONTRACTOR REPORT.** Scientific and technical findings by NASA-sponsored contractors and grantees.
- **CONFERENCE PUBLICATION.** Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.
- **SPECIAL PUBLICATION.** Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.
- **TECHNICAL TRANSLATION.** English-language translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services that complement the STI Program Office's diverse offerings include creating custom thesauri, building customized data bases, organizing and publishing research results . . . even providing videos.

For more information about the NASA STI Program Office, see the following:

- Access the NASA STI Program Home Page at <http://www.sti.nasa.gov>
- E-mail your question via the Internet to help@sti.nasa.gov
- Fax your question to the NASA Access Help Desk at 301-621-0134
- Telephone the NASA Access Help Desk at 301-621-0390
- Write to:
NASA Access Help Desk
NASA Center for AeroSpace Information
7121 Standard Drive
Hanover, MD 21076



Tribological Performance of Some Pennzane® Based Greases for Vacuum Applications

Mario Marchetti, William R. Jones, Jr., and Kenneth W. Street
Glenn Research Center, Cleveland, Ohio

Donald Wheeler, Duane Dixon, and Mark J. Jansen
Sest, Inc., Middleburg Hts., Ohio

Hiroshi Kimura
Kyodo Yushi Company, Ltd., Kanagawa, Japan

National Aeronautics and
Space Administration

Glenn Research Center

Acknowledgments

This work was performed while the author held a National Research Council Research Associateship Award in the Tribology and Surface Science Branch, NASA Glenn Research Center.

Trade names or manufacturers' names are used in this report for identification only. This usage does not constitute an official endorsement, either expressed or implied, by the National Aeronautics and Space Administration.

Available from

NASA Center for Aerospace Information
7121 Standard Drive
Hanover, MD 21076

National Technical Information Service
5285 Port Royal Road
Springfield, VA 22100

Available electronically at <http://gltrs.grc.nasa.gov/GLTRS>

Tribological Performance of Some Pennzane® Based Greases for Vacuum Applications

Mario Marchetti, William R. Jones, Jr., and Kenneth W. Street

National Aeronautics and Space Administration

Glenn Research Center

Cleveland, Ohio 44135

Donald Wheeler, Duane Dixon, and Mark J. Jansen

SEST Inc.,

Middleburg Hts., Ohio 44130

Hiroshi Kimura

Kyodo Yushi Co., Ltd.

Kanagawa, Japan 251-8588

Abstract

Commercial greases for space applications usually fulfill the requirements imposed by the severe conditions of use. The main requirement is their ability to create an EHL film, boundary film, or both under speed, load and temperature conditions that the mechanisms will operate. Three greases, all based on a multiply alkylated cyclopentane (Pennzane®) base oil, were studied. The thickeners were an n-octadecylterephthalate soap, a lithium soap, and a urea derivative. A Four-Ball Tribometer and a Spiral Orbit Tribometer were employed to evaluate the greases under ultrahigh vacuum. Results indicated that all three greases yielded very low wear rates and extended lifetimes. In addition, routine physical property data is reported for each grease.

I- Introduction

Extended mission lifetimes and improvements to other spacecraft components, such as electronics, batteries, and computers have placed increased burdens on space lubrication systems [1]. Fluid lubrication, either as a liquid or a grease, is commonly used to extend lifetimes and minimize wear, torque, and noise [2]. Because of these demands, the reliability of spacecraft moving mechanical assemblies (MMAs) clearly depends on the lubricant employed to cope with the increased lifetime. In order to assure that mission lifetimes will be completed, accelerated testing has become mandatory and critical.

Although full scale life testing [3] or actual component testing [4, 5, 6] is desirable, they are costly and time consuming. Various accelerated tests are available to evaluate the torque, wear rate, friction coefficient or degradation rate of the lubricant. These include the eccentric bearing test apparatus [7], the vacuum four-ball tribometer [8], and the spiral orbit rolling contact tribometer (SOT) [9].

The eccentric bearing apparatus employs an actual bearing in which an intentional misalignment is introduced to accelerate the degradation rate of the lubricant. Results with this device have correlated well with actual space experience [10].

The vacuum four ball tribometer consists of a rotating ball sliding against three stationary balls that are immersed in a lubricant, either liquid or grease. The wear rate of the

balls is calculated from the wear scars generated on the stationary balls. There is an inverse correlation between wear rate and component lifetimes [11].

The spiral orbit tribometer (SOT), fully described in a former study [9], reproduces the kinematics of an angular contact bearing. Data, such as the lubricated lifetime, friction coefficient, contact resistance, and degradation products can be determined, monitored, and analyzed. The relative lifetimes of lubricants measured with the SOT have correlated well with actual bearing life tests [4]. Previously limited to solid lubricants and liquid lubricants, the SOT's ability was recently extended to greases [12], which represent many of the lubricants used in MMAs.

The objective of this work was to compare the wear rates of steel and the lubricated lifetimes of three different vacuum greases based on Pennzane® using the vacuum four-ball apparatus and the spiral orbit tribometer. Each contained a different thickener: an n-octadecylterephthalate soap, a lithium soap, and a urea derivative. In addition, the terephthalate based grease contained an antiwear additive and two antioxidants.

II- Materials and Experimental Conditions

1- The Four-Ball Tribometer

This tribometer is used to test the lubricants' ability to reduce wear of bearing elements under high loads and operates with standard bearing balls. It is illustrated in Figure 1. The specimens are made of AISI 440C stainless steel. Balls were grade 25, 9.5 mm (3/8 inch) diameter. A 200 N load was applied through a pneumatic system (corresponding to an initial Hertz stress of 2.7 GPa). The top plate rotated and the sliding generated a wear scar on each stationary ball. The tests were conducted at a speed of 100 rpm. The test automatically began when the pressure dropped below $1.3 \cdot 10^{-4}$ Pa. The tests were performed at room temperature. The wear scars were measured after each hour with a microscope and the wear volume was calculated. This value is calculated considering the material worn has the shape of a spherical zone. Wear volume was plotted as a function of sliding distance and a wear rate calculated from a linear regression. At least four tests were run for each grease.

2- The Spiral Orbit Tribometer (SOT)

A Spiral Orbit Tribometer (SOT) simulates an angular contact bearing (Figure 2). A 12.7 mm (1/2 inch) ball rolled between a fixed plate and a rotary plate, running at 210 rpm. The load, providing a mean Hertz stress of 1.5 GPa, was applied through the fixed plate. The combination of the high load, the moderate speed, and of the small amount of lubricant (approximately 50 μ g) allowed the system to operate in the boundary lubrication regime. The ball was rolling and pivoting in a spiral and maintained in the orbit by the guide plate. The force the ball exerted on the guide plate was used to determine the friction coefficient, since the ball was sliding between the disks at this moment. The resistance of the contacts between the ball and the plates was calculated from the voltage drop across the plates. The evaluation of the greases was conducted at room temperature (≈ 23 °C), and under ultrahigh vacuum ($1.3 \cdot 10^{-6}$ Pa). As the lubricant was tribologically stressed, it was degraded and eventually consumed. Test conclusion was defined when a friction coefficient of 0.28 was attained. Normalized lubricant lifetime (or inversely, its degradation rate) was then defined as the number of orbits divided by the amount of lubricant in micrograms.

3- Materials preparation

The greases are all based on a multiply alkylated cyclopentane (Pennzane[®]) oil [13]. A summary of the composition and physical properties is provided in Table 1. Rheolube[®] 2000 is a commonly used grease for space applications, containing both anti-wear and anti-oxidant additives [14]. The grease MULTEMP 1C408 is manufactured with a lithium soap of 12-hydroxystearate (8% by weight), while the MULTEMP 1C409 contains a thickener made with a urea derivative (13% by weight). Neither MULTEMP grease contains any additives.

For the SOT tests, the greases were applied only on the ball by rolling it several times between two elastic membranes made of polyethylene. The small amount of grease deposited on the ball (30-60 μg) was determined using a balance with an accuracy of $\pm 2 \mu\text{g}$. The edges of the wear tracks on the SOT plates, where some of the deposited lubricant was pushed aside during the test, were analyzed with an infrared micro-spectrometer. It confirmed that both oil and thickener were present on the ball surface.

All specimens were made of AISI 440C stainless steel. For tribological purposes, ball and plate surfaces were polished to a roughness Ra of 0.05 μm . The parts were first rubbed with an alumina slurry and rinsed under running deionized water. Then they were ultrasonically cleaned for ten minutes each first in a bath of hexane, followed by deionized water. All drying was done with filtered nitrogen. The procedure was completed by exposing the specimens to UV/ozone for 15 minutes. More details can be found in reference [15]. In the case of the Four-Ball Tribometer, the cleaning baths were hexane, acetone and methanol, respectively (10 minutes each).

III- Results

1- Surface analysis

X-Ray Photoelectron Spectroscopy

An XPS analysis of some ball surfaces was conducted for the greases based on the lithium soap and urea derivative. The objective was to look for lithium and nitrogen traces on the surfaces after the tests. In both cases, no traces of lithium nor of nitrogen were detected. The ball surfaces had a slight brown coloration. The XPS spectra revealed the presence of a thick carbon layer, created by degradation of the greases. The thickness of this layer was great enough ($> 40 \text{ \AA}$) to obscure the iron substrate peaks. Moreover, in the case of lithium, the Li 1s peak is usually weak and interferes with one of the iron peaks. A similar interference problem occurred with an EDAX analysis of the same balls. This implies that the either Li-soap and urea derivative were completely consumed, or that the amounts left, either in its virgin state or after degradation, were too small to be detected.

Infrared Spectroscopy

An infrared analysis was made inside and adjacent to the wear tracks on the disks from the SOT (Figure 3). It confirmed the presence of organic degradation products (broad and weak bands) in the track and of the original grease on the edges of the tracks (sharp and strong bands) (Figure 4).

Raman Spectroscopy

Raman spectroscopy was conducted mainly in the scrub area on the bottom plate. The scrub (Figure 2) is the straight line portion of the orbit where the ball is sliding against the plates when it strikes the guide plate. Lubricant degradation mainly takes place in this area. Examples of this area are shown in Figure 3. The appearance of the sliding area is different between the greases. The degradation is more pronounced for the Rheolube® 2000 than for the other two greases. This was confirmed by the difficulty to detect a signature of degraded lubricant from MULTEMP 1C409 and any signal in the case of MULTEMP 1C408. A comparison of the Raman spectra for all the greases is presented in Figure 5. The spectra have a high fluorescence background. The ones for Rheolube® and MULTEMP 1C409 were obtained under the same conditions. No significant spectral characteristics of degradation products were detected with MULTEMP 1C408 after several attempts in different locations of the scrub. Hence, no comparable spectra were obtained on this grease.

Two broad peaks are evident with Rheolube® 2000, one centered near 1580 cm^{-1} and a smaller peak around 1360 cm^{-1} . These are related to the so called “G” and “D” peaks assigned to amorphous carbon [16], the final stage of the lubricant degradation. This has been observed in other friction polymers [17] and in an in-situ Raman study [18].

2- Wear rates

The wear rates of the steel for each grease (based on four tests) appear in Figure 6 along with reference data for Pennzane® 2001A base oil and formulated Pennzane® 2001. As can be seen, all greases yield low wear rates compared to fluorinated lubricants [8, 11]. These rates are only slightly higher than the formulated oil and base oil.

3- Friction coefficient and Lifetime in the SOT

At least four tests were conducted for each grease. All of them have large normalized lifetimes (number of orbits per microgram of lubricant) compared to fluorinated lubricants [9]. Results are shown in Figure 7. However, the lifetime of Rheolube® 2000 is several times greater than the MULTEMP 1C408 and 1C409 ones. The initial friction coefficients (Figure 8) are low in most cases (0.09-0.10), except for the MULTEMP 1C408 (0.12), but this value was stable during most of the test. For comparison, the initial friction coefficient for Pennzane® base oil is about 0.08.

The friction traces of the greases could be divided into two different behaviors. The first one is the grease with a urea-derivative thickener. The friction coefficient of this grease increased progressively and slowly until failure. The second one included the Li-soap and the Rheolube® 2000 greases. Both of these have also shown a progressive failure, but the increase in friction began at approximately half the lifetime. Thus, these two greases have shown a precursor of the failure, indicated by the arrows in Figure 8.

IV- Discussion

In general, wear rates from the vacuum four-ball tests have correlated well with the spiral orbit tribometer and full-scale bearing tests. That is, high wear rates of the steel are associated with short relative lifetimes of the lubricant (high degradation rates) and short

bearing lives. However, in this study, all greases yielded similar and low wear rates ($\approx 2.0 \cdot 10^{-10} \text{ mm}^3/\text{mm}$) compared to fluorinated lubricants [8, 11]. Fluorinated lubricants have generally yielded steel wear rates, at least one order of magnitude higher.

On the other hand, the relative lifetimes from the SOT show a clear distinction. The thickener made with sodium n-octadecylterephthalate soap has yielded the greatest lifetime (more than 15000 orbits/ μg). The greases evaluated use the same base oil. Therefore, the ability of these greases to lubricate a contact mainly depends on the ability of the grease to release the oil it contains. The oil is usually enclosed in a three dimensional network made by the thickener, the reaction to create the thickener sometimes taking directly place within the oil. So the oil is released when the grease is “squeezed” by the passage of the ball. This aspect was accurately detailed in reference [19]. Nevertheless, in spite of the evidence of the presence of both the oil and the thickener on the grease applied to the ball, it cannot be determined that the weight percentages are the ones of the original lubricant. It is also clear, from the Raman analysis, that the degradation of the MULTEMP greases has left very little chemical products. The only parameters that change are the presence of additives, the thickener nature or its weight percentage, and, as a consequence, the nature of the interactions between it and the oil.

The additives in Rheolube[®] 2000 generally have higher vapor pressures compared to the base oil. Therefore, even though “trapped” they are within the structure of the grease, they are not designed to stand ultrahigh vacuum.

The SOT operated with only 30 to 60 μg of grease. Assuming the lubricant is evenly distributed over the surface of the ball, the grease thickness is around 15 nm, while the surface roughness is $\text{Ra}=0.05 \mu\text{m}$. Considering the Stribeck criteria (lambda ratio, lubricant thickness/roughness), the tests were operated in the boundary regime. A resistance equal to zero confirmed this when the tests started, indicating direct contact between the plates and the ball surface. Neither inlet nor outlet were created and the Hamrock and Dowson model cannot be applied to calculate the film thickness. The only supply of lubricant was the initially charge present on the ball surface. Thus, it is clear that the grease composition and its structure will determine if the lubricant will be able to last. According to these results, the method that the small amount of oil present within the grease structure is released changed between the three lubricants evaluated. Also, the thickeners of the MULTEMP greases are not as capable as the one of the Rheolube[®] 2000 to cope with the severity of the conditions. This is consistent with the worked penetration data of the greases (Table 1). This test gives the depth of penetration of a cone falling in grease under defined conditions [19]. The MULTEMP 1C408 has the lowest one (Table 1), but also with the images and the spectroscopic analysis (Figures 3, 4 and 5) the grease leaving few residues. Grease 1C408 (Li-soap) and 1C409 (urea derivative) have not left residues on the wear tracks. The degradation of the MULTEMP thickeners occurred and lead to more volatile products. This would explain the shorter lifetimes and the lower amount of residue on the wear track.

Moreover, a change in the friction coefficient trace appeared in the case of grease MULTEMP 1C408 and Rheolube[®] 2000 at about half of the lifetime (see Figure 8). Therefore, we have a precursor of the failure, as observed before [12]. That implies that the degradation process of the lubricant (oil and thickener) is taking place later with the greases based on the ester soaps. However, the grease with a urea-derivative thickener has shown a constant increase in the friction coefficient, as did the neat oil. The consumption of the lubricant started immediately with the grease based on the urea derivative. The behavior of the urea grease can also be linked to the ability of the thickener to release the lubricant

within the contact. Thus, since the behavior of the MULTEMP 1C409 (urea derivative) was very close to that of the neat oil, we can assume that this grease quickly released the main part of the oil it contained, although the thickener percentage is higher, while the Rheolube® 2000 and the MULTEMP 1C408 (Li soap) greases released it more progressively. This could show that the interaction between the soap and the oil is stronger with the esters soaps.

It is also interesting to note the way the lubricant detected on the edge of the wear track behaves. The grease is supposed to provide oil from the edge to the tracks. But no oil was detected on the track after the conclusion of the test. It appears that the lubricant pushed out of the track is unavailable for the duration of the test. This aspect would reinforce the idea that a lubricant reservoir would still be necessary.

V- Conclusion

The greases based on Pennzane® oil have demonstrated a good ability to operate under severe conditions. The accelerated tests have shown that the grease based on the sodium n-octadecylterephthalamate soap has a greater lifetime than the ones with a lithium or a urea derivative thickeners. All these greases are able to provide good surface protection against wear and provide low friction coefficients. Without structural and rheological data, the effect of different interactions between the thickener and the oil to explain their tribological behaviors could not yet be confirmed. An increase in the thickener mass percentage of the Li-grease could improve its lifetime.

The Spiral Orbit Tribometer, an angular contact bearing simulator, has now clearly confirmed its ability to evaluate and optimize greases. It would also be able to establish a relationship between the amount of lubricant used and the lifetime of a mechanism.

References

1. Fleischauer P.D. and Hilton M.R., "Assessment of the Tribological Requirements of Advanced Spacecraft Mechanisms," The Aerospace Corporation, El Segundo, CA, Report n° TOF-0090 (5064)-1, 1991.
2. Jones W.R. Jr. and Jansen M.J., "Space Tribology," NASA/TM—2000-209924, March 2000.
3. Brown P.L., Miller J.B., Jones W.R. Jr., Rasmussen K., Wheeler D.R., Rana M., and Peri F., "The Clouds and Earth's Radiant Energy System Elevation Bearing Assembly Life Test," *Proceedings of the 33rd Space Mechanisms Symposium, May 1999*, pp. 197–212.
4. Loewenthal S.H., Jones W.R. Jr., and Predmore R., "Life of Pennzane and 815Z-Lubricated Instrument Bearings Cleaned with Non-CFC Solvents," NASA/TM-209392, 1999.
5. VanDyk S., Dietz B., Street K., Jones W., Jansen M., Dube M., Sharma R., and Predmore R., "The Role of Bearing and Scan Mechanism Life Testing in Flight Qualification of The MODIS Instrument," *Proceedings of the 35th Aerospace Mechanisms Symposium, May 2001*, pp. 1–14.
6. Gill S. and Rowntree R.A., "Interim Results from ESTL Studies on Static Adhesion and the Performance of Pennzane SHF X-2000 in Ball Bearings," *Proceedings of the 6th European Space Mechanisms and Tribology Symposium, Technopark, Zürich, Switzerland, 4–6 October 1995*, pp. 279–291.

7. Kalogeras C., Hilton M., Carré D., Didziulis S., and Fleischauer P., "The Use of Screening Tests in Spacecraft Lubricant Evaluation," The Aerospace Corporation, El Segundo, CA, Report n° TR-93(3935)-6, 1993.
8. Masuko M., Jones W.R. Jr., Jansen R., Ebihara B., Pepper S.V., and Helmick L.R., "A Vacuum Four-Ball Tribometer to Evaluate Liquid Lubricants for Space Applications," *Lubrication Engineering*, volume 50, n°11, pp. 871–875.
9. Jones W.R. Jr., Pepper S.V., Jansen M.J., Nguyen Q.N., Kingsbury E.P., Loewenthal S., and Predmore R., "A New Apparatus to Evaluate Lubricants for Space Applications—The Spiral Orbit Tribometer (SOT)," *International Spring Fuels and Lubricants Meeting and Exposition*, Paris (France), June 19–22, 2000.
10. Carré D., Kalogeras C.G., Didziulis S.V., Fleischauer P.D., and Bauer R., "Recent Experience with Synthetic Hydrocarbon Lubricants for Spacecraft Applications," The Aerospace Corporation, El Segundo, CA, Report n° TR-95(5935)-3, 1995.
11. Jones W.R. Jr., Poslowski A.K., Shogrin B.A., Herrera-Fierro P., and Jansen M.J., "Evaluation of Several Space Lubricants Using a Vacuum Four-Ball Tribometer," NASA/TM-208654, 1998.
12. Marchetti M., Jones W.R. Jr., Street K.W., Pepper S.V., and Jansen M.J., "Preliminary Evaluation of Greases for Space Mechanisms Using a Vacuum Spiral Orbit Tribometer," *NLGI 68th Conference*, West Palm Beach (Florida), October 27–31 2001.
13. Venier C.G., Casserly E.W., and Gunsel S., "Tris (2-Octyldodecyl) Cyclopentane, a Low Volatility, Wide Liquid Range, Hydrocarbon Fluids," *8th International Colloquium of Tribology "Tribology 2000", Esslingen (Germany), 14–16 January 1992*. article n° 13.1, p. 12.
14. Space Tribology Handbook, edited by European Space Tribology Laboratory (s.l.), 1997.
15. Jansen M.J., Jones W.R. Jr., and Wheeler D.R., "Evaluation of Non-Ozone-Depleting-Chemical Cleaning Methods for Space Mechanisms Using a Vacuum Spiral Orbit Rolling Contact Tribometer," NASA/TM—2000-210050, May 2000.
16. Dillon R.O., Wollam J.A., and Katkanant V., "Use of Raman Scattering to Investigate Disorder and Crystallite Formation in As-Deposited and Annealed Carbon Films," *Phys. Rev. B*, 29, 3482, 1984.
17. Herrera-Fierro P., Shogrin B.A., Jones W.R. Jr., "Spectroscopic Analysis of Perfluoroether Lubricant Degradation During Boundary Lubrication," NASA TM-107299, 1996.
18. U C.C. and Stair P.C., "In situ Study of Multialkylated Cyclopentane and Perfluoropolyalkyl Ether Chemistry in Concentrated Contacts Using Ultraviolet Raman Spectroscopy," *Tribol. Lett.*, volume 4, pp. 183–170, 1998.
19. Courronné I., "Etude Expérimentale du Comportement de Graisses Lubrifiantes pour Roulements à Billes," Thesis, INSA Lyon, France (00 ISAL 0083), 1st December 2000.

	GREASES		
	Rheolube® 2000	MULTEMP 1C408	MULTEMP 1C409
Appearance	light brown	light brown	light brown
Additive(s)	a phosphate, an amine and a hindered phenol	none	none
Thickener	soap of sodium n-octadecylterephthalamate	soap of lithium 12-hydroxy stearate	urea derivative
Mass percentage of thickener	≈ 15	≈ 8	≈ 13
Dropping point (°C)	> 260	209	260
Worked penetration, 60 strikes	276	227	300
Oil separation (100 °C, 24 h) (mass %)	3.3	1.1	1.7

Table 1: Greases compositions and properties

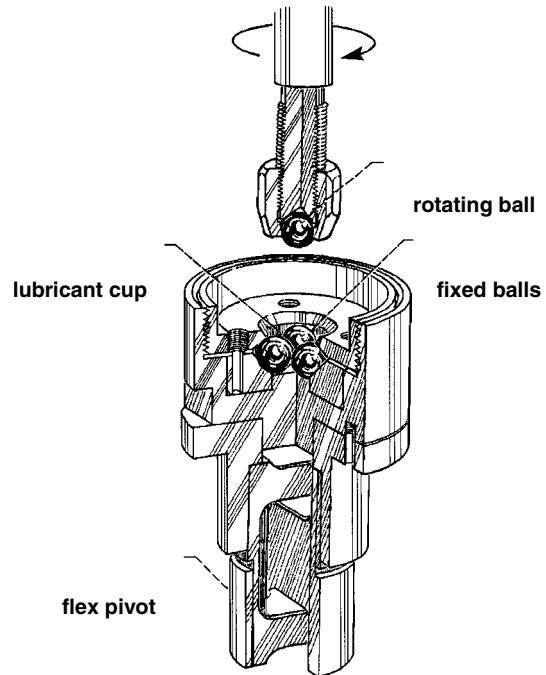


Figure 1: The Four-Ball Tribometer

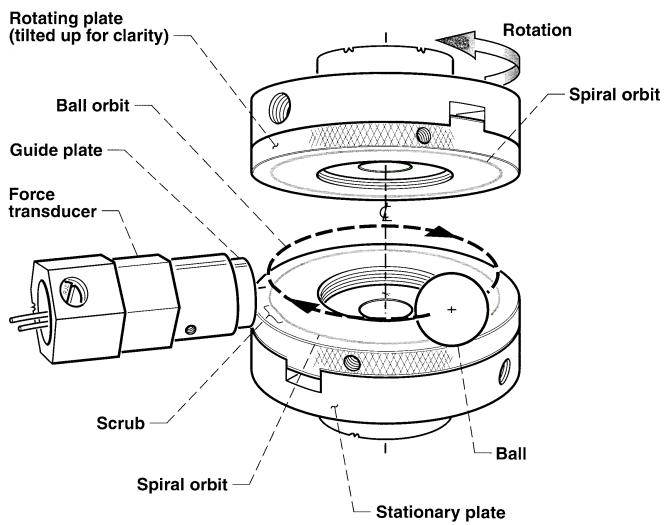


Figure 2: The Spiral Orbit Tribometer

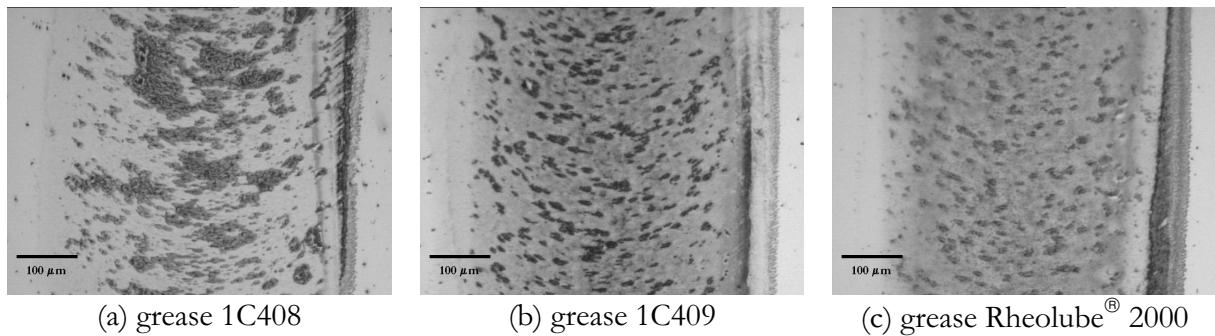


Figure 3: Pictures of the scrub area for the three Pennzane® based greases

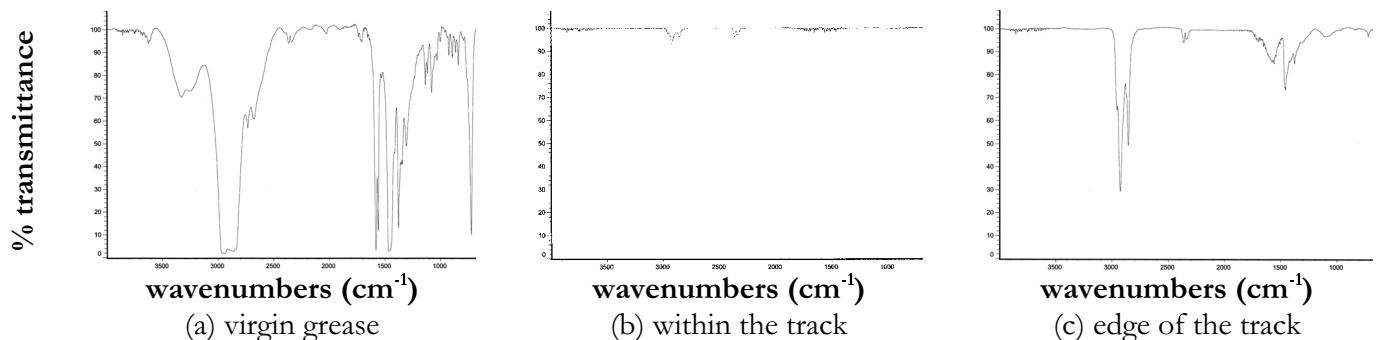


Figure 4: Examples of infrared signatures of the grease 1C408

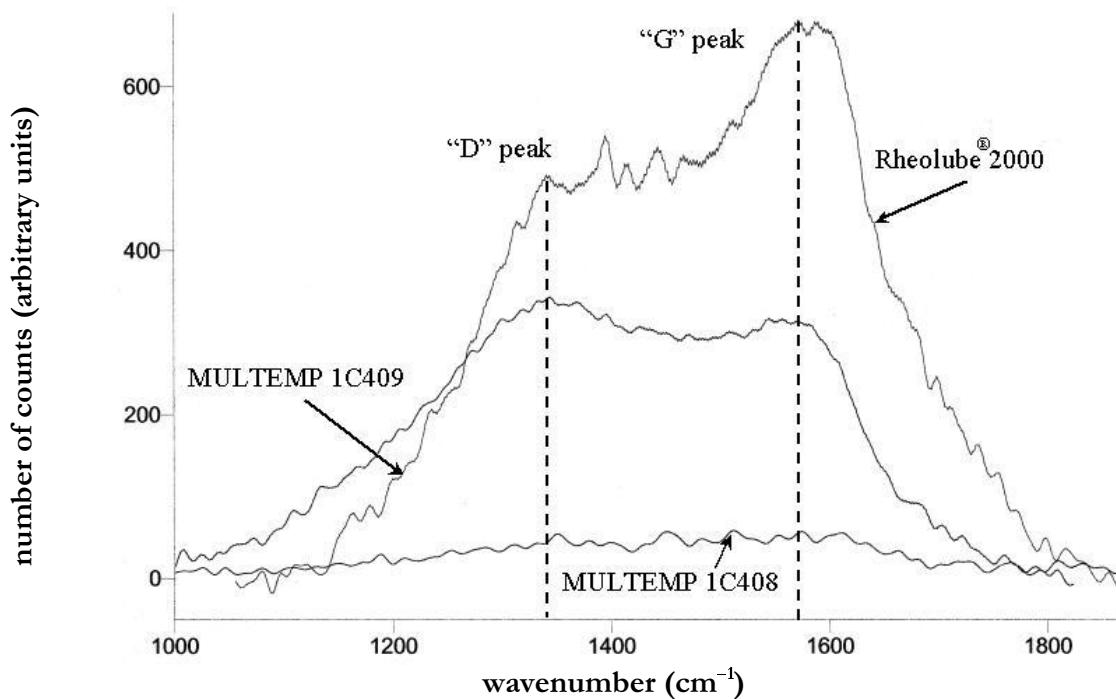


Figure 5: Raman spectra of the scrub area for the different Pennzane® based greases

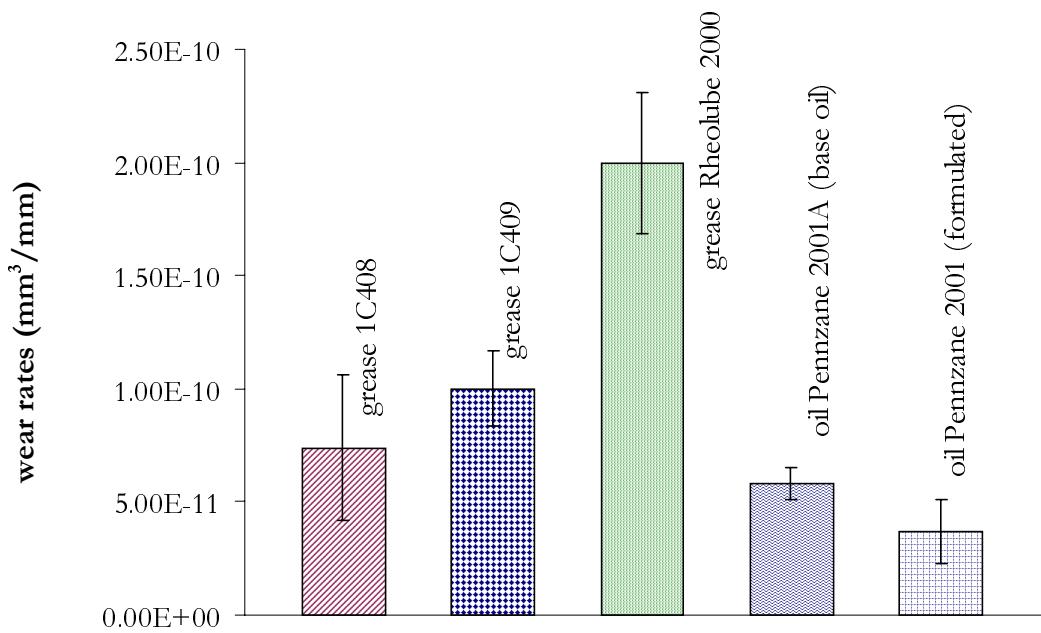


Figure 6: Wear rates (mm^3/mm) of the different Pennzane® oils and greases

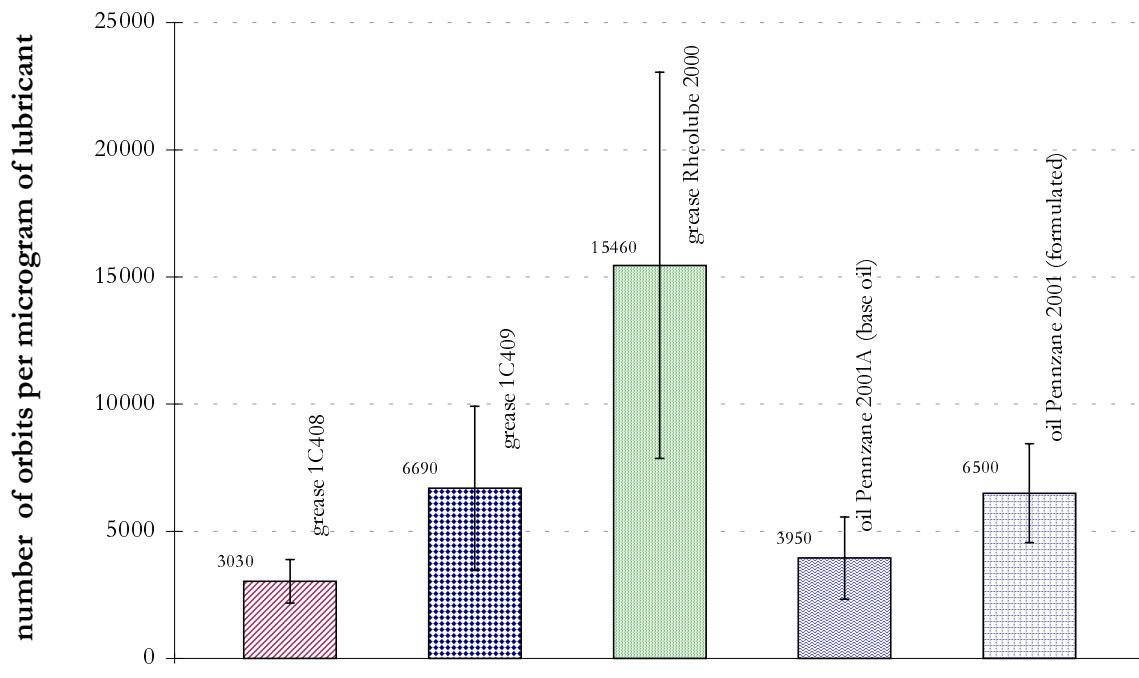


Figure 7: Normalized lifetimes of the Pennzane® greases and oils (with standard deviation for the four tests)

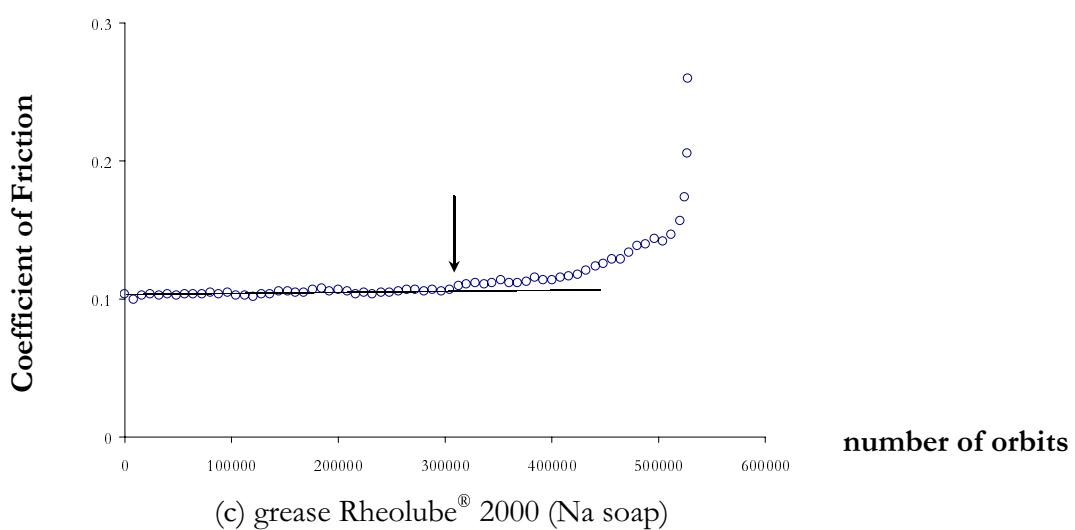
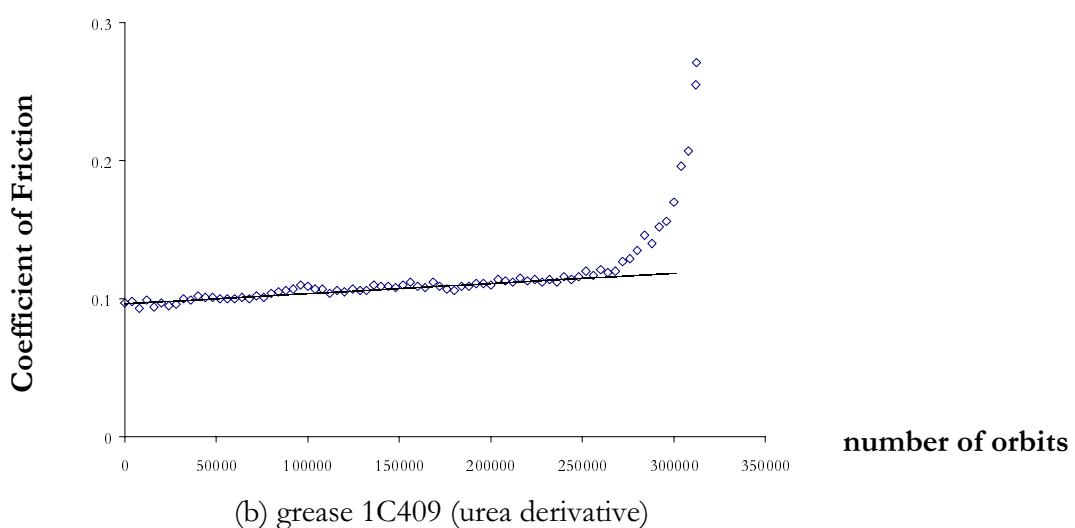
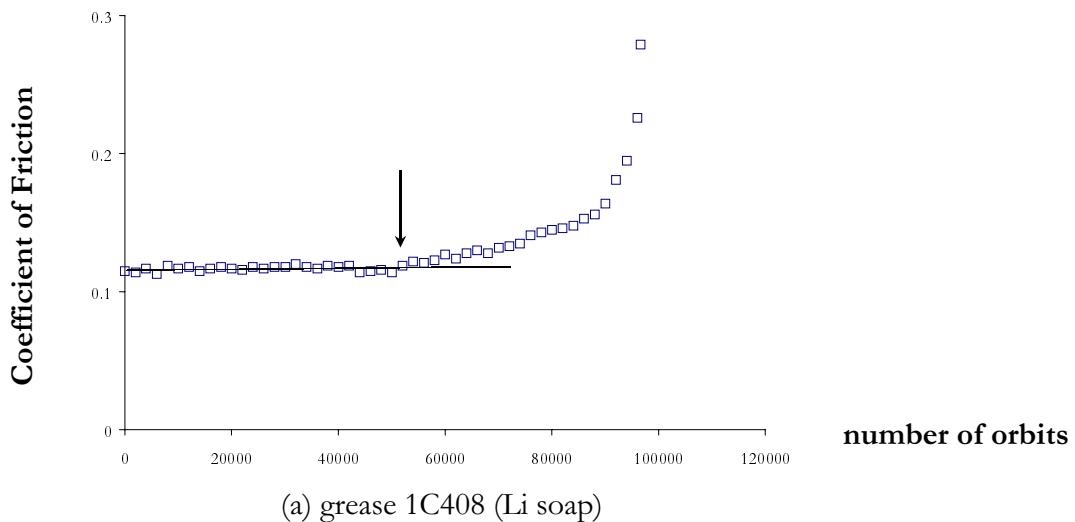


Figure 8: Examples of friction coefficient traces

REPORT DOCUMENTATION PAGE			<i>Form Approved OMB No. 0704-0188</i>
<p>Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.</p>			
1. AGENCY USE ONLY (Leave blank)	2. REPORT DATE	3. REPORT TYPE AND DATES COVERED	
	February 2002	Technical Memorandum	
4. TITLE AND SUBTITLE		5. FUNDING NUMBERS	
Tribological Performance of Some Pennzane® Based Greases for Vacuum Applications		WU-274-00-00-00	
6. AUTHOR(S)		7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)	
Mario Marchetti, William R. Jones, Jr., Kenneth W. Street, Donald Wheeler, Duane Dixon, Mark J. Jansen, Hiroshi Kimura		National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-3191	
8. PERFORMING ORGANIZATION REPORT NUMBER		9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)	
E-13198		National Aeronautics and Space Administration Washington, DC 20546-0001	
10. SPONSORING/MONITORING AGENCY REPORT NUMBER		11. SUPPLEMENTARY NOTES	
NASA TM—2002-211373		<p>Mario Marchetti, William R. Jones, Jr., and Kenneth W. Street, NASA Glenn Research Center; Donald Wheeler, Duane Dixon, and Mark J. Jansen, <i>Sest, Inc.</i>, Middleburg Hts., Ohio; Hiroshi Kimura, Kyodo Yushi Company, Ltd., Kanagawa, Japan. Responsible person, M. Marchetti, organization code 5960, 216-433-5843.</p>	
12a. DISTRIBUTION/AVAILABILITY STATEMENT		12b. DISTRIBUTION CODE	
Unclassified - Unlimited Subject Category: 27		Distribution: Nonstandard Available electronically at http://gltrs.grc.nasa.gov/GLTRS This publication is available from the NASA Center for AeroSpace Information, 301-621-0390.	
13. ABSTRACT (Maximum 200 words)			
<p>Commercial greases for space applications usually fulfill the requirements imposed by the severe conditions of use. The main requirement is their ability to create an EHL film, boundary film, or both under speed, load and temperature conditions that the mechanisms will operate. Three greases, all based on a multiply alkylated cyclopentane (Pennzane®) base oil, were studied. The thickeners were an n-octadecylterephthalate soap, a lithium soap, and a urea derivative. A Four-Ball Tribometer and a Spiral Orbit Tribometer were employed to evaluate the greases under ultrahigh vacuum. Results indicated that all three greases yielded very low wear rates and extended lifetimes. In addition, routine physical property data is reported for each grease.</p>			
14. SUBJECT TERMS			15. NUMBER OF PAGES
Grease lubrication; Space mechanisms; Accelerated tests			18
			16. PRICE CODE
17. SECURITY CLASSIFICATION OF REPORT	18. SECURITY CLASSIFICATION OF THIS PAGE	19. SECURITY CLASSIFICATION OF ABSTRACT	20. LIMITATION OF ABSTRACT
Unclassified	Unclassified	Unclassified	